

Numerical simulation of dispersion around a cubic building: characterization of wind-induced pressure coefficients on cube using Standard K- ϵ Model

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Abstract:

The aim of the work is to determine the total pressure coefficient (C_p), the behavior of pressure and pressure coefficient C_p near the walls of cube (roof and floor of cube) to study the effect of C_p on the behavior of flow and reverse flow on the top of cube due to wind tunnel, which allows the pressure due to the wind at the cube faces (**negative pressure** ($C_p < 0$) or **overpressure** ($C_p > 0$) compared to the **static pressure** (atmospheric pressure)) to be **quantified**. This coefficient depends on the wind velocity (**W**) upstream the cube (velocity in the wind tunnel), on the **geometry** of obstacle (which is cube used here) and on the attack **angle** of wind.

Keywords: K- ϵ Model, Pressure Coefficient, Wind (flow), CFD, Fluent.

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1. Introduction:

CFD (RANS)simulations results of this work were compared with The experimental wind-induced mean pressure coefficients from **Martinuzzi et al. (1993) [3]** on “the flow around surface-mounted, prismatic obstacles placed in a fully developed channel flow” and **Castro and A. G. Robins (1977, England) [8]** on their study “*The flow around a surface-mounted cube in uniform and turbulent streams*”, in order to validate the CFD modeling approach, which could be used for predicting pressure coefficients on other geometries. One wind directions was considered (Zero degree) in the direction of horizontal axis . All of the simulations were performed with the commercial ANSYS FLUENT 14.5 software (www.ansys.com) [5] using Standard **K- ϵ** Model.

It should be noted that the modeling is consistent to the Recommendation of **Martinuzzi (1993, 1997) [3]** and **Mohd ARIFF, 2009 [2]**. **Ricciardi, Thomas, Sandrine (2015, France) [9]** based on their study : “*Experimental and numerical characterization of wind-induced pressure coefficients on nuclear buildings and chimney exhausts*” studied more cases related to Pressure coefficient fields C_p on buildings for various wind directions ; and concluded in the results that areas of over-pressure can be clearly observed on the windward faces ($C_p > 0$) and areas of negative pressure ($C_p < 0$) on the other faces (leeward faces , roof) based on a n experimental study and RANS simulations using SST k- ω turbulence model performed on most cases of wind directions.

Castro and A. G. Robins (1977, England) [8] on their study “*The flow around a surface-mounted cube in uniform and turbulent streams*” studied An experimental investigation of the flow around surface-mounted cubes that confirm the conclusion of **Ricciardi, Thomas, Sandrine** about distribution of coefficient pressure C_p (where $C_p < 0$ and $C_p > 0$) over different regions of cube as will be shown below in the results.

Also **Ricciardi, Thomas, Sandrine (2015, France)** recommended that the range of C_p values obtained for a null air flow in the chimney (null air flow (V/U) = 0), noted as $C_{pV/U=0}$, **characterizes the aspiration capacity** of

the chimney. If this range of $C_{p_{V/U=0}}$ is **entirely negative**, it indicates that, whatever the wind direction, the wind will **create an aspiration in the chimney** and then **will assist the exhaust of the air**. If, for some wind directions, the values of $C_{p_{V/U=0}}$ are **positive**, the wind can cause **back-pressure** on the top of the chimney (or cube) and then thwart the exhaust of the air. Where \mathbf{V} is air flow from chimney (or from a vent exist in the center of roof cube), and \mathbf{U} is the wind speed at any point around cube.

This study can help us more to identify the range of coefficient pressure (C_p) of a solid cube ($\mathbf{V}/\mathbf{U}=0$). In this case (**solid cube**), no air flow will be investigated from the roof of cube ($\mathbf{V}/\mathbf{U}=0$) as will be shown in the results below *Fig 5,6,7*.

2. Steps of Modeling and setting up of CFD details using Ansys Fluent (14.5) :

2.1 Importing and Scaling the grid (meshing): Actually, the first important step in **Ansys Fluent** (14.5) is how to generate the best mesh (grid), as will be shown below.

2.2 Importing and Scaling the grid :

First, the **numerical solver** is set to the following specifications:

- Solver type: **Pressure based.**
- Formulation: **Implicit**
- Space: **3D**
- Time: **steady** (we want here **steady** not transient).
- Velocity formulation: **Absolute**
- Gradient option: **Green-Gauss cell based.**

2.3 Selecting the physical model:

The turbulence-model used is: **standard k-ε (skε) [1]**:

It is a model based on model transport equations for the turbulence kinetic energy (k) and its dissipation rate (ϵ). The model transport equation for k is derived from the exact equation, while the model transport equation for ϵ was obtained using physical reasoning and bears little resemblance to its mathematically exact counterpart. In the derivation of the k - ϵ model, the assumption is that the flow is **fully turbulent**, and the effects of molecular viscosity are **negligible**. The **standard k-ε model** is therefore valid only for **fully turbulent flows as shown**:

The turbulence kinetic energy, k , and its rate of dissipation, ϵ , are obtained from the following transport equations:

$$\frac{\partial}{\partial t}(\rho k) + \frac{\partial}{\partial x_i}(\rho k u_i) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + G_k + G_b - \rho \epsilon - Y_M + S_k$$

and

$$\frac{\partial}{\partial t}(\rho \epsilon) + \frac{\partial}{\partial x_i}(\rho \epsilon u_i) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_\epsilon} \right) \frac{\partial \epsilon}{\partial x_j} \right] + C_{1\epsilon} \frac{\epsilon}{k} (G_k + C_{3\epsilon} G_b) - C_{2\epsilon} \rho \frac{\epsilon^2}{k} + S_\epsilon$$

Where:

G_k : represents the generation of turbulence kinetic energy due to the mean velocity gradients, calculated as described in Modeling Turbulent Production in the k - ϵ Models.

G_b : is the generation of turbulence kinetic energy due to buoyancy, calculated as described in effects of buoyancy on turbulence in the k- ϵ Models.

Y_M : represents the contribution of the fluctuating dilatation in compressible turbulence to the overall dissipation rate, calculated as described in Effects of Compressibility on Turbulence in the k- ϵ Models.

$C_{1\epsilon}, C_{2\epsilon}, C_\mu$ are constants with following default values are: **1.4, 1.92, and 0.09** respectively, and $\sigma_k, \sigma_\epsilon$ with following default values: **1.0, 1.3** respectively. S_k, S_ϵ are user-defined source terms. The turbulent (or eddy) viscosity, μ_t , is computed by combining k and ϵ as follows; where, C_μ is a constant :

$$\mu_t = \rho C_\mu \frac{k^2}{\epsilon}$$

2.4 Prescribing operating conditions:

The operating pressure is set to normal **atmospheric pressure (101325 Pa)**. The effect of gravity is incorporated in the model by setting the **gravity value** equal to **-9.8 m/S²** along **y-axis**.

2.5 Applying boundary conditions:

Following boundary conditions are applied to the above-defined zones (Named-selection):

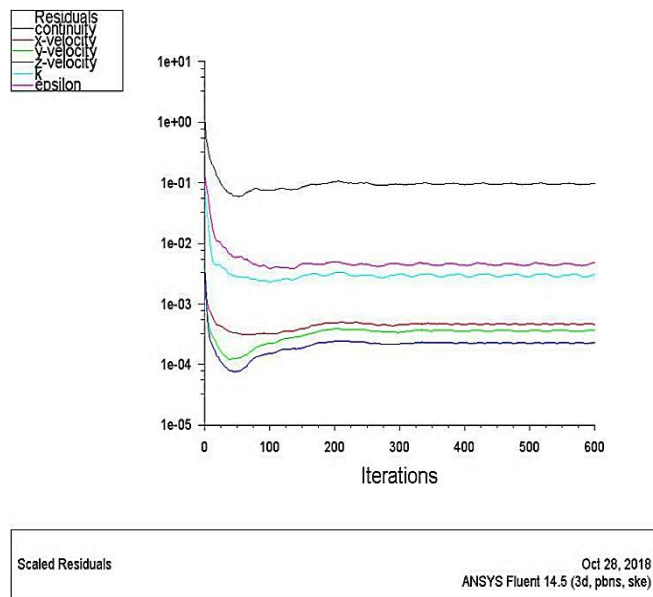
- Setting “**velocity inlet**” as 0.6 m/s with direction x-coordinate (uniform velocity as **absolute**).
- Setting “**outlet pressure**” as 0 Pa.
- Setting all sides of cube : as “ **wall**” with neglecting the slip along these walls.
- Set the top side of domain once as “**symmetry**” and once as “**wall**”.

2.6 Setting up solver controls:

The solver controls are set to the following settings:

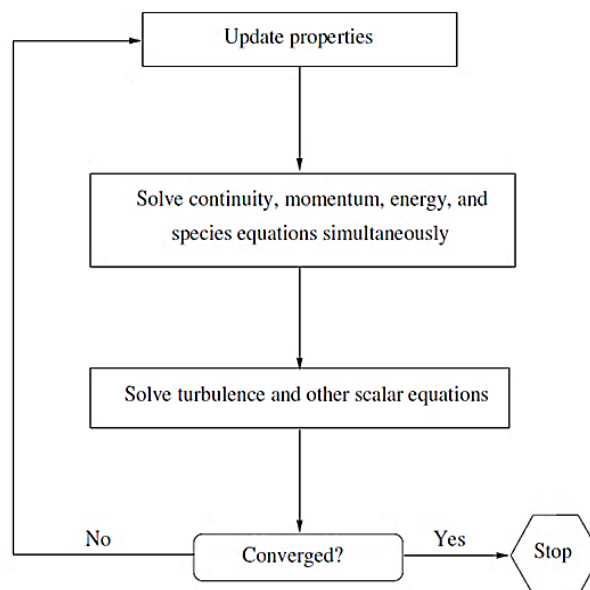
- *Equations: **Flow and Energy***
- *Pressure-velocity coupling: **SIMPLE***
- *Under-Relaxation factors:*
 - *Pressure = **0.3***
 - *Density = **1***
 - *Body Forces = **1***
 - *Momentum = **0.7***
 - *Turbulence Kinetic Energy = **0.8***
- *Discretization:*
 - *Pressure = **PRESTO!***
 - *Momentum = **Second Order Upwind***
 - *Energy = **Second Order Upwind***

2.7 Setting up convergence monitors: The convergence monitors are set appropriately and the plotting of residuals (**continuity, x-velocity, y-velocity, z-velocity,....etc**) during the calculations have been enabled as shown:



2.8 Initializing the flow field:

The calculations were started by giving the number of iterations with standard initialization from the inlet velocity **0.6 (m/s)**.



2.9 Computing and Monitoring the Solution:

The discretized conservation equations are **solved iteratively**. A number of iterations are usually required to reach a **converged** solution as shown above.

Convergence is reached when:

1. Changes in solution variables from one iteration to the next are negligible. Residuals provide mechanism to help monitor this trend.
2. Overall property conservation is achieved.

3. Numerical Method

3.1 Governing equations:

The equation for **conservation of mass, or continuity** equation, can be written as follows:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \vec{U}_i) = S_m$$

This equation is the general form of the mass conservation equation and is valid for **incompressible** as well as **compressible** flows. The source S_m is the mass added to the continuous phase from the dispersed second phase (e.g., due to vaporization of liquid droplets) and any user-defined sources.

-The mass conservation equation (for steady-state, incompressible, single phase flow) used in this study:

$$\rho (\nabla U_i) = 0 \dots\dots\dots (1)$$

- **Conservation of momentum** in an inertial (non-accelerating) reference frame is described by :

$$\frac{\partial}{\partial t} (\rho \vec{U}_i) + \nabla (\rho \vec{U}_j \vec{U}_i) = -\nabla P + \nabla \cdot (\vec{\tau}) + \rho \vec{g} + \vec{F}$$

Where P is the static pressure, $\vec{\tau}$ is the stress tensor (described below), and $(\rho \vec{g})$ and \vec{F} are the gravitational body force and external body forces (e.g., that arise from interaction with the dispersed phase), respectively. \vec{F} also contains other model-dependent source terms such as porous-media and user-defined sources. The stress tensor $\vec{\tau}$ is given by:

$$\vec{\tau} = \mu [(\nabla \vec{U}_i + \nabla \vec{U}_i^T) - \frac{2}{3} \nabla \cdot (\vec{U}_i \mathbf{I})]$$

Where μ is the molecular viscosity, \mathbf{I} is the unit tensor, and the second term on the right hand side is the effect of volume dilation.

- The momentum conservation equation (for steady-state, incompressible, single phase flow):

$$\nabla (\rho \vec{U}_j \vec{U}_i) = -\nabla P + \nabla \cdot (\vec{\tau}) + \rho \vec{g} \dots\dots\dots (2)$$

Where i is the index for the x, y, z direction, $\rho (\frac{kg}{m^3})$ is the density of the fluid, P is the static pressure ($\frac{N}{m^2}$), $\mu (\frac{kg}{m.s})$ is the viscosity coefficient, which is sum of the dynamic viscosity μ_d and fluid turbulent viscosity and μ_t

- μ is the total viscosity coefficient:

$$\mu = \mu_d + \mu_t \dots\dots\dots (3)$$

Where, $\mu_t = \rho C_\mu \frac{k^2}{\epsilon}$

3.2 CFD Setting up of cube closure:

The computational domain ($14.5H \times 2H \times 9H$) in X , Y and Z respectively shown in **Fig 1 , 2 and 4** is identical to that used by **Mohd ARIFF, Salim M.(2009) [2]** , **Lakehal and Rodi (1997)[6]** in their numerical analysis that replicates the experimental setup of **Martinuzzi et al. (1993)**. A fully developed turbulent flow was set at the inlet and $Re_H = 40,000$. No-slip conditions were applied on the channel floor, top wall and all cube faces, whereas the side walls were defined as symmetry to reduce computational cost, since they are sufficiently far from the cube to influence the flow characteristics **Fig 3**.

RANS simulations using Standard $k-\epsilon$ turbulence model were performed on all of the study using coarse grid generated using ANSYS 14.5 software with 300,000 (coarse mesh of $Y+ > 33$) elements of Geometry as seen in **figure 3**.

The mean local pressure on a face due to the wind P_j is linked to the mean reference dynamic pressure of the wind, noted as $(\frac{1}{2} \rho W_{ref}^2)$, and the mean pressure coefficient C_{pj} is defined by the following relation (4) and (5):

$$C_p = \frac{P - P_{atm}}{\frac{1}{2} \rho W_{ref}^2} \dots\dots\dots (4)$$

$$C_{pj} = \frac{P_j - P_0}{\frac{1}{2} \rho W_{ref}^2} \dots\dots\dots (5)$$

Where:

P_j : mean local pressure due to the wind at a point j (Pa),

P_0 : static pressure far from the building influence zone (Pa),

ρ : air density (kg/m³),

W_{ref} : wind reference velocity, measured at reference height (which is here inflow wind velocity) (m/s).

C_{pj} : mean pressure coefficient at a point j.

(Non dimensionless)

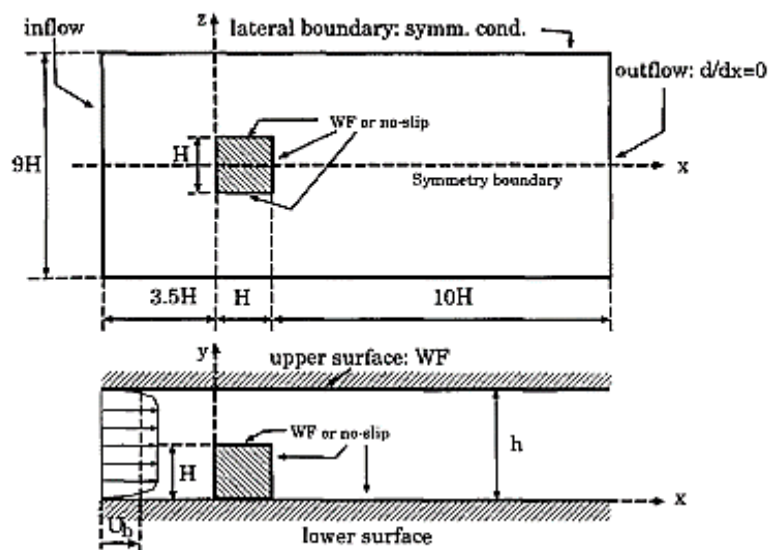


Fig1. Geometry with boundary conditions [6]

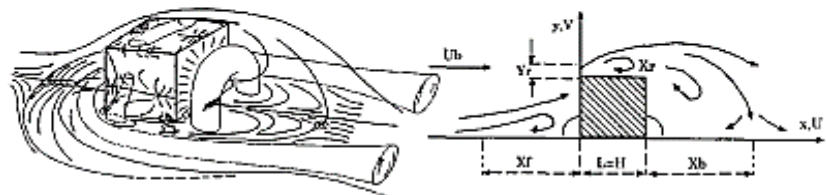


Fig2. Representation of flow around a cube [6]

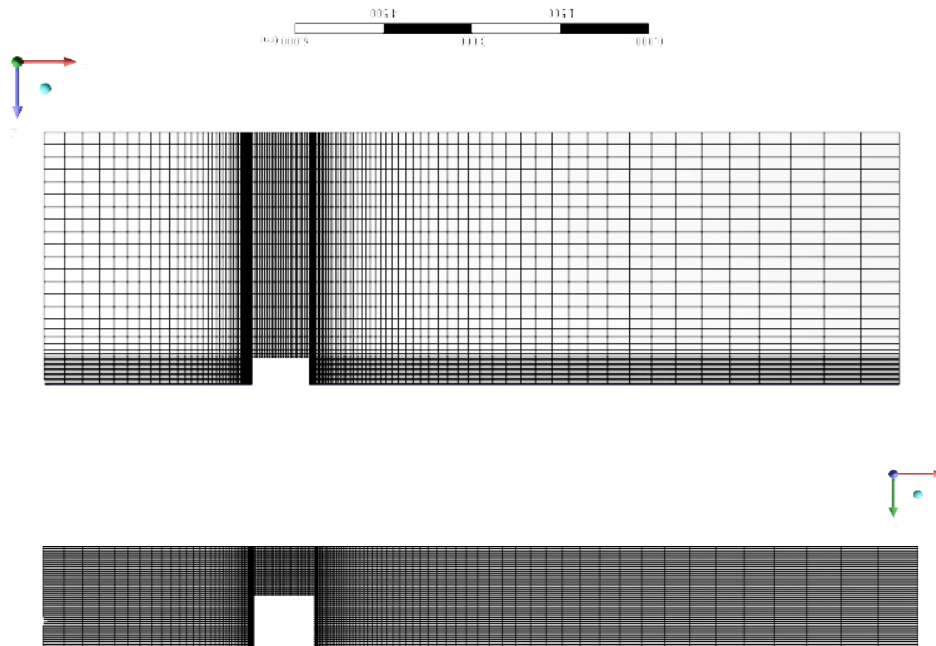


Fig 3. Generated Mesh in Ansys 14.5

Velocity inlet is **0.6 m/s**, pressure outlet is atmospheric pressure put to **0**, Reynold number is almost **40000**, the number of elements are about **300,000 elements (Y+>33)**, with neglecting slipping on the sides of walls.

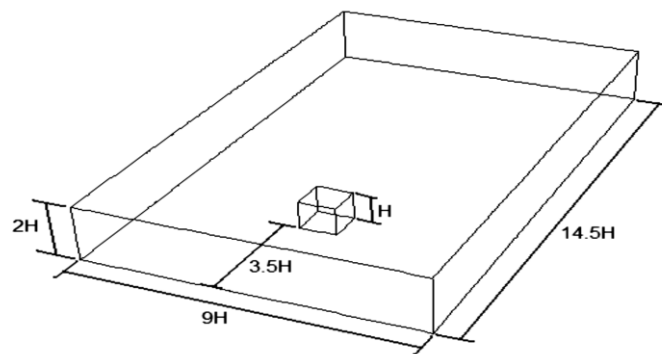


Fig4. Geometry of cube closure

4. Results:

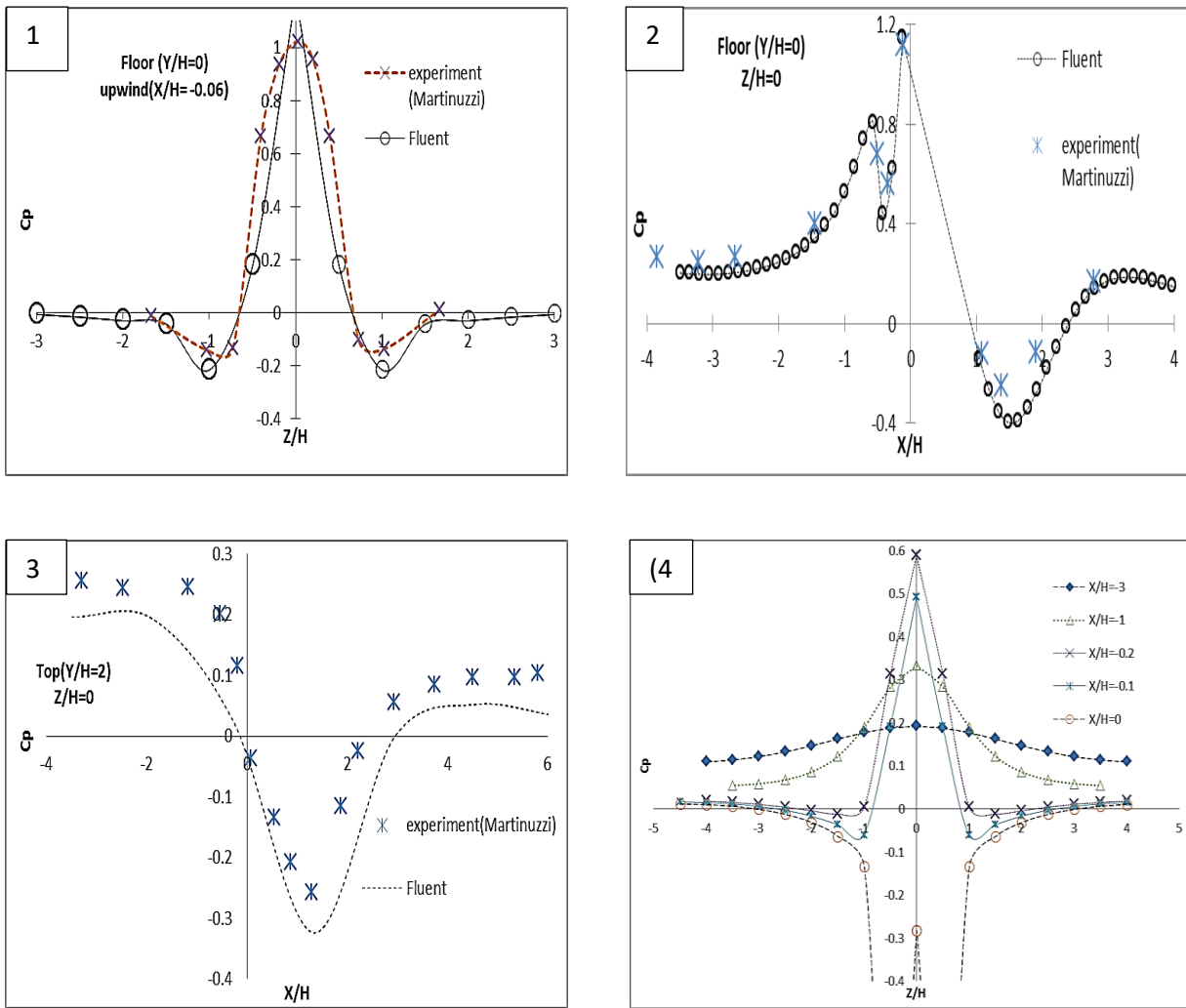


Fig 5. Comparison between the Pressure coefficient of present work and experimental data from Martinuzzi (1993) : (1) in plane $Y/H=0$ & Line $X/H = -0.06$ (2) in planes $Y/H=0$ & Line $Z/H=0$; (3) in planes $Y/H=2$ & Line $Z/H=0$; (4) at Lines : $X/H= -3$, $X/H= -1$, $X/H= -0.2$, $X/H= -0.1$, $X/H= 0$ on the floor ($Y/H=0$)

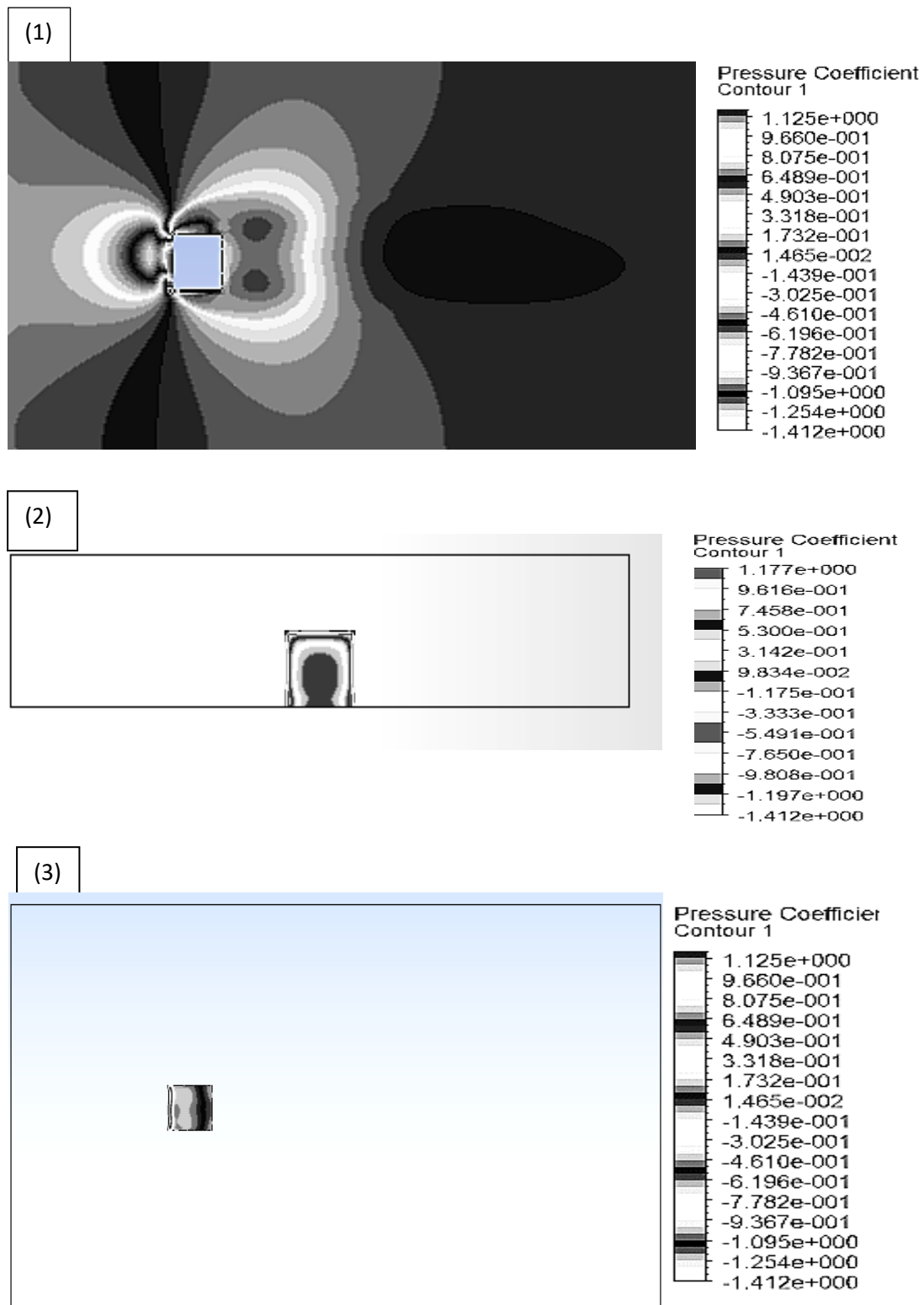


Fig 6. Coefficient pressure contours on: (1) the floor plane ($Y/H=0$). (2) the upwind side of cube ($X/H=0$). (3) the top side of cube

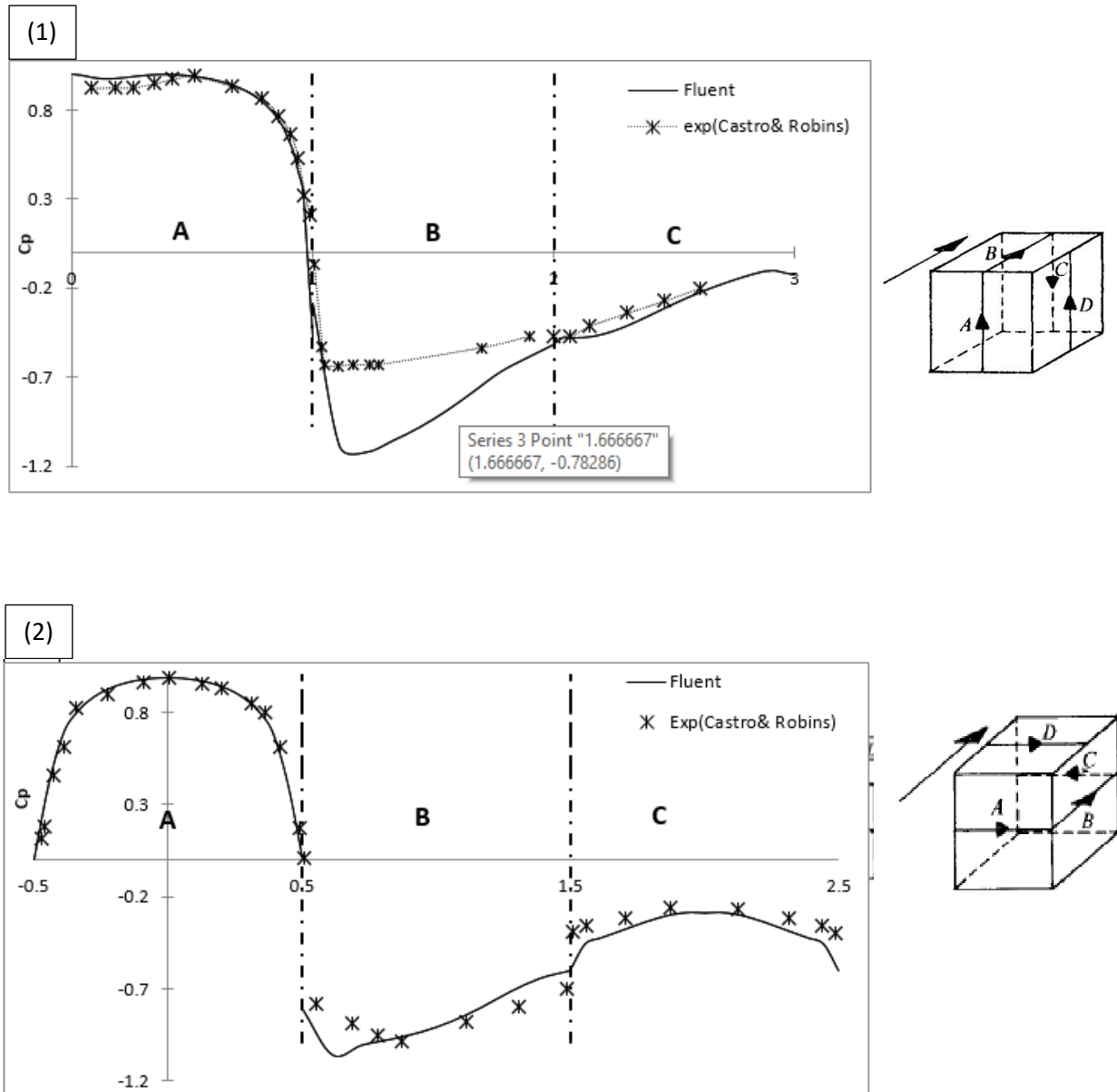


Fig7. Coefficient pressure on the different sides of cube closure with different directions as shown in (1) longitudinal direction (2) lateral direction with a comparison from Castro & Robins experiment.

5. Discussion and Conclusion:

All the results of this study are based on Steady RANS simulation in ANSYS Fluent 14.5 and we have got a good acceptance with experimental works. **Fig5** represent a comparison between the Pressure coefficient of present work and experimental data from Martinuzzi on the floor and top of cube; where **Fig 5.1** illustrates the behavior of coefficient pressure around the cube on the floor along the lateral (z-axis) with range **(-0.2:1)**, the maximum value of C_p occurs at center line of cube ($Z=0$) on the floor, the minimum occurs in the symmetrical sides of geometry at points ($Z=-1$ & $Z=1$). **Fig 5.2** illustrates the behavior of coefficient pressure around the cube on the floor along the horizontal (x-axis) at center line of cube with range **(-0.4:1.1)**, the maximum value of C_p occurs at a point before ($x=0$) from the upwind side on the floor.

Fig 5.3 illustrates the behavior of coefficient pressure around the cube on the top of cube along the horizontal center line (x-axis) with range **(-0.3:0.2)**, the minimum value of C_p occurs at point before ($X=1$) of value **-0.3**, and the **fig 5.4** describes the behavior of C_p on the floor along the lateral (Z-axis); how it is changing with increasing the maximum point of C_p at $Z=0$ in the upwind region gradually from the farther upwind distance line ($X=-3$) approaching to the line distance ($X=0$) with sudden change of **high negative value of C_p** . Contours of C_p on: (1) the floor in plane ($Y/H=0$). (2) the upwind side of cube at plane. ($X/H=0$). (3) the top side of cube show the behavior of pressure coefficient changes. (**fig 6.1**, **fig 6.2** and **fig 6.3** respectively).

Due to presence the cube in the domain of wind tunnel, velocity of wind will be affected to perform the reverse flow near the roof of the cube as shown in **fig2**; where velocity profiles start to change near the cube forming the shown behavior along the horizontal direction.

Also, we can notice that the range of coefficient Pressure C_p on the face **A** with directions as shown in **fig 7.1** and **fig 7.2** is **entirely positive** ($C_p > 0$), Also, the range of coefficient Pressure C_p on the faces **B,C** of solid cube in the different directions is **entirely negative** ($C_p < 0$). As shown in **fig 7.1** and **fig 7.2**.

Nomenclature:

H - Height of Obstacle (m).	Re_H - Reynolds Number (= $(H * u_B) / \nu_{air}$).
Z - crosswind direction (m).	U - Instantaneous Velocity (m/s).
X - Horizontal Distance along Stream wise direction (m).	XF - Frontal Separation length (H).
Y - Vertical Distance normal to the ground direction.	k - Turbulent kinetic energy (m^2/s^2).
u_B - Bulk Velocity (m/s).	u_τ - Friction Velocity (m/s).
XR - Reattachment Length (H).	ε - Dissipation rate (m^2/s^2).
y+ - Dimensionless distance to the wall.	ν_{air} - Kinematic viscosity of air (m^2/S).

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